

PCT Appl. No. : **PCT/NO00/00191**

Filed : **Herewith**

**IN THE SPECIFICATION:**

**A1**  
Page 1, immediately after the title, please insert **Related Applications** This application claims the benefit of the Norwegian applications 1999 2739 filed June 4, 1999 and 1999 5140 filed October 21, 1999 and the international application PCT/NO00/00191 filed June 2, 2000. This application is related to co-pending applications "TRANSLATION AND LOCKING MECHANISM IN A MISSILE" serial number \_\_\_\_\_, attorney docket number PROTEC7.001APC, "RETARDING AND LOCK APPARATUS AND METHOD FOR RETARDATION AND INTERLOCKING OF ELEMENTS" serial number \_\_\_\_\_, attorney docket PROTEC8.001APC, and "PROPELLING DEVICE FOR A PROJECTILE IN A MISSILE" serial number \_\_\_\_\_, attorney docket number PROTEC9.001APC, all filed concurrently herewith. **Background of the Invention Field of the Invention--**

**A2**  
Page 1, line 6, please insert **Description of the Related Art--**

**A2**  
Page 1, line 25, please insert **Summary of the Invention--**

**A3**  
Page 2, line 25, please insert **Brief Description of the Drawings--**

**A4**  
Page 3, immediately before line 1, please insert **Detailed Description of the Preferred Embodiments--**

**IN THE CLAIMS:**

Please amend the Claims as follows:

**A5**  
1. (Amended) A release mechanism between a projectile and a rocket motor in a missile, where the projectile is released from the rocket motor during the flight of the missile when the rocket motor is burned out and retardation occurs, wherein the rocket motor in the front end thereof comprises a forward closure, one in the forward closure received and movable lock, at least one lock, at least one spring that biases against the lock retainer in a direction opposite to the direction of travel for the missile, and wherein the projectile in the rear end thereof has a central boss surrounded by the forward closure of the rocket motor, wherein the